Equation (11) should read

$$\frac{p(0)}{T(0)\rho_g C_{pg}} < < \frac{u_g}{R\beta}$$

The relationship after Eq. (11) should read

$$\frac{p(0)}{T(0)\rho_g C_{pg}} = \dots$$

Equation (15) should read $\tau_p > > \tau_d \times 10^{-3}$. The lines in Fig. 1 are now: $\tau_p = 10^{-6}/p$, $\tau_p = 3 \times 10^{-3}p$ and

Thus, the quasi-steady domain is defined as follows:

for
$$p \le 1.1 \times 10^{-4}$$
, $\tau_p > > 10^{-6}/p$

for
$$1.1 \times 10^{-4} \le p \le 3$$
, $\tau_p > 9 \times 10^{-3}$

for
$$p \ge 3$$
,

$$\tau_n > 3 \times 10^{-3} p$$

Technique for Determining Local Heat-Transfer Coefficients

H. J. Sternfeld and J. Reinkenhof DFVLR, Lampoldshausen, Germany [AIAA J. 15, 105-109 (1977)]

R QUATION (11) should read

$$\alpha = -\sum_{(i)} \sum_{(j)} (a_{ij}ix_0^{i-1}t^j)$$

$$\frac{\sum_{(\mu)} [b_{\mu}^{i} (\sum_{(i)} \sum_{(j)} a_{ij} x_{0}^{i} t^{j})^{\mu}]}{\sum_{(k)} a_{k} t^{k} - \sum_{(i)} \sum_{(j)} a_{ij} x_{0}^{i} t^{j}}$$
(11)

Equation (12) should read

$$A_{\theta} \alpha \left(T_{\delta} - T_{w} \right) - A_{\theta} \left(\frac{x_{I}}{x_{\theta}} \right)^{\epsilon} \left[-\lambda \frac{\partial T}{\partial x} \right]_{x = x_{I}}$$

$$= \frac{d}{dt} \int_{x_{\theta}}^{x_{I}} A_{\theta} \left(\frac{x}{x_{\theta}} \right)^{\epsilon} \rho c T \, dx \tag{12}$$

Equation (13) should read

$$\alpha(t) = \frac{A - B}{C}$$

$$\min \{t_n\} \le t \le \max \{t_n\} \tag{13}$$

Equation (14) should read

$$A = \frac{\mathrm{d}}{\mathrm{d}t} \int_{x_0}^{x_f} \rho c T \, x^{\epsilon} \mathrm{d}x \tag{14}$$

Acoustic Thermometric Measurements of Propellant Gas Temperature in Guns

E.M. Schmidt, E.J. Gion, and D.D. Shear U.S. Army Ballistic Research Laboratory, Aberdeen Proving Ground, Md.

[AIAA J, 15, 222-226 (1977)]

I N the footnote on page 222, the correct paper number for the paper presented at the AIAA/SAE 12th Propulsion Conference is AIAA Paper 76-643.

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Index categories: Combustion in Heterogeneous Media; Nozzle and Channel Flow; Reactive Flows.

Structure of Turbulent Shear Flows: A New Look

A. Roshko

California Institute of Technology, Pasadena, Calif. [AIAA J., 14, 1349-1357(1976)]

N p. 1355, second column, line 8, reference 31 should be changed to reference 34; on line 49, reference 37 should be 36; and on line 63, reference 38 should be 37. On p. 1356, first column, line 4, references 39, 41 should be 39, 40, and 41; on line 36, legth should be length. On p. 1353, second column, line 9, it should read y/x = -0.095. On p. 1354, first column, line 4, 1-M should read UM. On Fig. 2, second line, L_1L/μ_1 should read U_1L/μ_1 .

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Index categories: Heat Conduction; Nozzle and Channel Flow; Liquid Rocket Engines.

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Index categories: Jets, Wakes, and Viscid-Inviscid Flow Interactions; Boundary Layers and Convective Heat Transfer-Turbulent.